Motivation	Previous Work	Materials and Methods	Results	Conclusions/Future Work
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Ignition of *n*-Hexane-Air Mixtures by Moving Hot Spheres

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24th International Colloquium on the Dynamics of Explosions and Reactive Systems

National Central University Taipei, Taiwan July 28 - August 2, 2013



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 - Accidental Ignition
 - Hot Surface Ignition
- 2. Previous Work
- 3. Materials and Methods
- 4. Results
- 5. Conclusions/Future Work

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Results Conclusions/Future Work

Accidental Ignition

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- $\rightarrow\,$ electrostatic ignition of fuel
- \rightarrow lightning strike
- \rightarrow electrical faults in pumps, fuel quantity instrumentation
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- Characterize fuel-oxidizer properties (n-hexane)
 - → ignition delay time (Burcat et al. and Zhukov et al.)
 - → heating rate on the low temperature oxidation of hexane by air (Boettcher et al.)
 - → minimum ignition temperature (Boettcher)
 - → minimum ignition energy (Bane)
 - \rightarrow laminar burning speed



TWA 800, NY 747-100, July 17, 1996



China Air Flight 120 caught fire in Okinawa Japan (BBC News, August 20, 2007)

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Hot Surfa	ace Ignition			

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- Lightning attaching to conductor

- Current flows through composite material
- Heating of material
- Ejection of hot particles (V_{particle} > 0)

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Ignition of *n*-hexane-air mixtures using moving hot particles

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- Results

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Stationary particles heated via IR laser : Beyer et al. (2010), Dubaniewicz et al. (2000), Homan et al. (1981) Moving heated particles introduced into a flammable environment : Silver (1937)



Experimental Setup, Beyer et al. (2010)



Experimental Setup, Silver (1937)





Pentane/air ignition results, Beyer et al. (2010) 3 vol-% pentane/air ignition, Silver (1937)

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Sphere diameter (mm)

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Pentane/air ignition results, Beyer et al. (2010) 3 vol-% pentane/air ignition, Silver (1937)

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1.5

2.5

3.5

vol-% pentane

4.5





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 $T_{stationary} < T_{moving}$ $D_{sphere} \uparrow T_{ignition} \downarrow$

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- Parallel flow past a vertical plate :
 - $ightarrow U_{\infty}$ = 0 m/s and U_{∞} = 2.5 m/s $ightarrow T_{\infty}$ = 300 K
- Adiabatic plate : $T_w = 1200 \text{ K}$





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Experimental Setup : Combustion Vessel





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 Experimental Setup : Combustion Vessel

11.7 cm diameter windows





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Experimental Setup : Combustion Vessel

11.7 cm diameter windows





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pressure manometer

Experimental Setup : Combustion Vessel





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pressure manometer

Experimental Setup : Combustion Vessel





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Experimental Setup : Combustion Vessel



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Experimental Setup : Combustion Vessel





- Observe changes in the density gradient of the fluid due to variations in the refractive index
- Visualize flame :
 - → very hot flame propagating into cold unburned reactants
- High speed camera :
 - ightarrow 10,000 frames per seconc ightarrow 512imes512 resolution





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Experime	ntal Setup :	High Current Pai	rticle Hea	ter



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 - Heating in Air
 - Ignition Results
 - Simulation Results
 - Comparison
- 5. Conclusions/Future Work

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Materials and Methods

Results : High Current Heating in Air

Heating in air, R= 17 m $\Omega,\,d=$ 4 mm, heating time \approx 1 sec



Titanium (Ti-6Al-4V) $\rightarrow T_{max} \approx 1200$ K



Titanium (Ti-6Al-4V) $\rightarrow T_{max} \approx 1300$ K

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Results : High Current Heating in Air

Heating in air, R= 13 m $\Omega,\,d=$ 4 mm



Material	$T_{melting}$ (K)
Copper	1350
Carbon steel, alloy steel, 302 SS	1700-1800
Titanium	1950

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n-Hexane–air : $P_0 = 100$ kPa, $T_0 = 300$ K, $\Phi = 0.9$ d = 4 mm, material : titanium, $V \approx 2.3$ m/s



 $T_{surface} = 1199 + 42$ K/-10 K



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Results : Flame Geometry

n-Hexane–Air, $\Phi = 0.9$ P = 100 kPa



 $V_S^0 =$ 2.6 m/s $V_P \approx$ 2.3 m/s

n-Hexane–O₂–40% N₂, $\Phi = 0.9$ P = 50 kPa



 $V_S^0 =$ 23 m/s $V_P \approx$ 2.3 m/s

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Simulation Setup

- Grid
 - \rightarrow Axisymmetric
 - $\rightarrow\,$ Locally refined around stationary hot particle
 - \rightarrow 192x128 mesh points
- Boundary conditions
 - $\rightarrow~$ Constant particle temperature
 - \rightarrow Wall temperature : 300 K
 - \rightarrow Inert surfaces
 - → Neumann boundary condition for species
- Initial conditions
 - $ightarrow P_0 =$ 100 kPa, $T_0 =$ 300 K and $\Phi =$ 0.9
 - \rightarrow Flow N₂ at t = 0-200 ms and *n*-heptane¹-air at t > 200 ms

 1 Blanquart G., Pepiot-Desjardins P., and Pitsch H. (2009). Combustion and Flame. 156 : 588-607.



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 Results : Simulations (N2 Hot Wake)
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 ${\rm N}_2$ velocity and thermal boundary layers, stabilized at t> 100 ms, $T_{surface}=$ 1500 K



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 $T_{surface} = 1500 \ {\rm K}$



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n-Heptane–air : $P_0 = 100 \text{ kPa}$, $T_0 = 300 \text{ K}$, $\Phi = 0.9 d = 4 \text{ mm}$, V = 2.5 m/s, $T_{surface} = 1450 \text{ K}$



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Results : Ignition Simulations

$T_{surface} =$ 1450 K, fuel mass fraction and temperature



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Results : Ignition Simulations

$T_{surface} =$ 1450 K, N_2 mass fraction and OH*



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Results : Experiments and Simulations



Ignition delay time

- Ignition delay time pprox 5 ms
- Increase in ignition delay time while approaching minimum ignition temperature

Experiment : $T_{surface} = 1199$ K, Simulation : $T_{surface} = 1450$ K



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| Conclusions/Future Work | | | | | | | |

Conclusions

- Ignition of *n*-hexane-air mixtures using moving heated particles ($P_0 = 100$ kPa, $T_0 = 300$ K, $\Phi = 0.9$) with $V_P \approx 2.3$ m/s
 - $\rightarrow \,$ 50% probability of ignition with $T_{surface} =$ 1170 K using 4 mm diameter titanium spheres
 - $\rightarrow\,$ Interaction of flame and sphere for mixtures with flame speeds comparable to the particle velocity
- Qualitative agreement between experiments and simulation
 - $\rightarrow\,$ Nitrogen recirculation region that leads to a particular flame propagation behaviour

- $\rightarrow\,$ Ignition close to sphere surface near flow separation region
- Comparable ignition delay times between experiments and simulations
 - ightarrow 35-60% difference at $T_{surface}$ > 1400 K

Future Work

- New particle heating method : CO₂ laser
 - \rightarrow Repeatable surface temperatures
 - ightarrow Use of metallic or non-metallic materials
 - $\rightarrow~$ Easy switch to larger or smaller particles

Motivation	Previous Work	Materials and Methods	Results	Conclusions/Future Work
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The present work was carried out in the Explosion Dynamics Laboratory of the California Institute of Technology and was supported by The Boeing Company through a Strategic Research and Development Relationship Agreement CT-BA-GTA-1.

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Thank You



David Marxer photography

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